The procurement of two (2) ILS/DME systems Suvarnabhumi International Airport /Runway 02L and Runway 20R Date June 13, 2025

SCOPE OF SPECIFICATIONS

1. TECHNICAL SPECIFICATIONS

2. CONSTRUCTION WORKS

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The procurement of two (2) ILS/DME systems Suvarnabhumi International Airport /Runway O2L and Runway 20R

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	SECTION 2 : CONSTRUCTION WORKS	
ลำดับ	รายชื่อเอกสาร	จำนวน
໑	ข้อกำหนดทั่วไป	
ම	ข้อกำหนดการปฏิบัติงานในพื้นที่ปลอดภัยรอบทางวิ่ง เขตการบิน และพื้นที่บริเวณท่าอากาศยาน	ด ฉบับ
ຄາ	ขอบเขตของงานก่อสร้างฯ หรืองานปรับปรุงฯ	໑ ຊນັນ
¢	แบบรูปงานก่อสร้างๆ และรายละเอียดประกอบแบบๆ	ය ຊນັບ
હે	รายการคำนวณโครงสร้างเกี่ยวกับสถานี Localizer และ Glide Path	๒ ຊບັບ
G	บัญชีแสดงรายการ ปริมาณงาน และราคา	໑ ຊບັບ

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SECTION 1

TECHNICAL SPECIFICATIONS

In the event that the proposed technical specifications are inconsistent with the published technical specifications on the manufacturer's website, AEROTHAI reserves the right to adhere to the information on the website.

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1. Definitions

[CO] In the specifications, the following words and expression shall have the meanings assigned to them here under except where the context otherwise requires :

AEROTHAI	Aeronautical Radio of Thailand Ltd.
Tenderer	The juristic person, firm, or company that offers to provide materials, or perform a service, or do a job for AEROTHAI at a specified cost or rate.
Contractor	The juristic person, firm or company whose tender(s)/proposal(s) has/have been accepted by AEROTHAI and who agrees to accomplish the activities for AEROTHAI.
Proposal	The response to the requirement specified in Scope of Specifications.
ICAO Annex 10, Vol. I	Aeronautical Telecommunications : Volume I, Radio Navigation Aids. Eighth Edition, July 2023, Amendment 93.
ICAO Doc 8071, Vol. I	Manual on Testing of Radio Navigation Aids : Volume I, Testing of Ground-based Radio Navigation Systems, Fifth Edition – 2018.
ICAO Annex 14, Vol. I	Aerodromes : Volume I, Aerodrome Design and Operations, Eighth Edition, July 2018, Amendment 15.
ICAO Doc 9157	Aerodrome Design Manual Part 6 : Frangibility, First Edition – 2006, Amendment 1.
FAA Order 6750.16E	Siting Criteria for Instrument Landing Systems, April 10, 2014.
FAA Order 8200.1D	United States Standard Flight Inspection Manual (USSFIM) with CHG 1, April 2015.
[CO] Comply Only	This symbol indicates that the tenderer shall comply and implement accordingly, with no supporting evidence required at the time of Tender.
[ET] Evidence with Tender	This symbol indicates that the Tenderer must provide relevant supporting evidence with the tender.
[EC] Evidence at Contract Stage	This symbol indicates requirements for which the Contractor must submit supporting evidence for approval during contract execution. The Tenderer is only required to acknowledge these items; no submission is required at the tendering stage.

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2. General Requirements

2.1	Two (2)	Two (2) complete systems of ILS/DME (Instrument Landing System /Distance Measuring Equipment)							
[CO]	are required at the following airport /runway (see also APPENDIX A) :								
	ltem	tem Airport /Runway							
	2.1.1	SUVARNABHUMI Airport /Runway 02L							
	2.1.2	SUVARNA	BHUMI Airpo	ort /Runway 20R					
2.2	For eac	h ILS/DME	system, the	e Tenderer shall propose th	ne equipme	ent, as follo	ows :		
[CO]	2.2.1	LOC, GP	and DME e	quipment, including their	respective	e antenna :	systems		
	[CO]	2.2.1.1	Localizer	(LOC) equipment shall co	onsist of du	ual two-fre	quency (2F)		
		[ET]	transmitte	rs and dual monitors.					
		2.2.1.2	Glide path	n (GP) equipment shall co	onsist of d	ual two-fre	quency (2F)		
		[ET]	transmitte	rs and dual monitors.					
		2.2.1.3	Distance M	leasuring Equipment (DME)	shall consi	st of dual t	transponders		
		[ET]	and dual r	nonitors.					
			2.2.1.3.1	DME equipment shall be DV	IE/N (Narrow	Spectrum Ch	naracteristics).		
			[ET]						
			2.2.1.3.2	DME equipment shall be	e capable d	of transmitt	ting at least		
			[ET]	100 watts of power - the	so called	"low-power	red" DME.		
			2.2.1.3.3	DME equipment shall be	e collocate	d with GP	equipment.		
			[ET]	However, the identification	on of DME	shall be sy	nchronized		
				with the identification o	of LOC.				
	2.2.2	Monitoring	Monitoring and control system, computers and network equipment are as specified in						
	[CO]	[6] [7] an	[6] [7] and [8], respectively.						
[CO]	The equ	uipment sha	all be finalize	ed and <u>not</u> in prototype forr	m at the tim	e the Tend	erer submits		
	the proposal.								
2.3	For eac	h ILS/DME	system, the	Tenderer shall design the	specification	is of LOC, C	GP and DME		
[ET]	equipm	nent and :	submit a s	upporting document (see	e APPENDI	X B), whic	h contains		
	simulat	ion results,	at least, as	follows :					
	Item	Parameter	rameter			GP	DME		
	2.3.1	Radiation	Patterns		1	1	1		
		(Require on	ly the case of	flat terrain without obstacles)	V	V	V		
	2.3.2	Structure			1	1			
		with respe	ect to LOC o	ourse or GP path	V	V	-		
							4		
							Are		

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	2.3.3	DDM Characteristics	1	1	10			
		with respect to LOC azimuth or GP elevation	V	V				
	2.3.4	Coverage (Usable Distance)	1	1	1			
		Power Density or Field Strength			v			
	Remark							
	"fi th an (Al Th co In co	or the simulation results of [2.3.1], the Tenderer lat terrain without obstacles" scenario, but for e Tenderer shall submit the results of both "flat-t d "terrain and obstacle" scenario which uses the PPENDIX C) as software inputs. The obstacles include <u>not</u> only buildings but also pordinates if such aircrafts affect the radiated guid case that the "terrain and obstacle" data provide mpatible with the simulation software, the Te o a format that is usable by the software, prov	those of [2.: errain withou data from " bositioned a ance (naviga ded by AERC nderer may	3.2] [2.3.3] ut obstacle Site Surve ircrafts at t tion) signal OTHAI are <u>r</u> convert	and [2.3.4], e" scenario <i>y Reports</i> " the specific <u>not</u> directly those data			
	 such as obstacle geometry and terrain profiles is preserved. 2. The simulation results of the proposed ILS/DME system shall demonstrate that suitable for <i>"Site Survey Reports"</i> (APPENDIX C) and also meets flight requirements and tolerances as stated in [Doc 8071 / Table I-4-7, Table I-4-8 and Tab 3-3 respectively] (see also [3.3.2] [3.3.3] [3.3.4] and [3.3.5] for LOC equipment, and [4.3.2] [4.3.3] [4.3.4] and [4.3.5] for GP equipment, and 							
		[5.3.2] and [5.3.3] for DME equipment)						
2.4 [CO]	constru	ntractor shall provide the LOC shelters and GP/I ction described in Section 2 : Construction Work Iters based on Frangibility Standard [Doc 9157, Part	s. AEROTHA	I has alread	dy designed			
3	ltem	Airport /Approach Runway	LOC Shel	ter GP/I	OME Shelter			
	2.4.1	SUVARNABHUMI Airport /Runway 02L	1		1			
	2.4.2	SUVARNABHUMI Airport /Runway 20R	1		1			
	Remark : The symbol " ✓ " marks the places that the Contractor shall provide LOC shelters and GP/DME shelters.							

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2.5 The Tenderer shall design, and who has become the the Contractor shall then provide [ET]the associated antenna supporters -- namely, "LOC Antenna Supporter" and/or "GP Antenna Tower" (including complete construction), which shall comply with Frangibility Standard [ANNEX 14 / Vol. I / Paragraph 9.9.3] and [Doc 9157, Part 6 - Frangibility / Chapter 5 or 6]. AEROTHAI will provide all necessary "Site Survey Reports" (APPENDIX C), including the runway profile, to support the Tenderer's design process. If the LOC antenna height exceeds 2 meters and the Tenderer considers that an LOC elevated platform is unnecessary, the LOC antenna supporters may alternatively be a "TOWER" type (Figure 2-1b), and shall be easy to climb and safe for maintenance activities. Figure 2-1 : Example of LOC antenna supporters, which the antenna height exceeds 2 meters a) LOC Elevated Platform b) LOC Support Tower Item Airport /Runway LOC Ant Supporter GP Ant Tower 2.5.1 SUVARNABHUMI Airport /Runway 02L as designed as designed 2.5.2 SUVARNABHUMI Airport /Runway 20R as designed as designed Remark : 1. Only if the runway profile causes the line-of-sight coverage problem between the antenna radiating element and threshold crossing height (TCH), which makes aircraft receiver unable to capture the "COURSE (CRS)" energy, the LOC antenna array may be installed on an elevated platform [FAA Order 6750.16E / Chapter 2 / 5. Siting Requirements / e. Elevation].

2. The height of LOC antenna shall be designed to be as low as possible; however, the simulation results of the coverage (usable distance) shall still meet flight test requirements and tolerances [2.3.4].

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The procurement of two (2) ILS/DME systems Suvarnabhumi International Airport /Runway 02L and Runway 20R

2.6	Genera	al requiren	nents for LOC, GP and DME equipment.						
[CO]									
	[CO]	2.6.1.1 Each unit of LOC, GP and DME equipment shall be designed to operate							
		[ET] on a single-phase AC power system with a nominal volta							
8		used in Thailand (220 VAC / 50 Hz).							
		2.6.1.2	Each unit of LOC, GP and DME equipment shall b	pe equipped with					
		[ET]	dual " <u>BUILT-IN</u> " AC/DC power supply modules, so t	hat no transmitte					
			is shut down when a failure occurs in one of the AC	/DC power supply					
			modules.						
		2.6.1.3	Each unit of LOC, GP and DME equipment shall b						
		[ET]	" <u>BATTERY</u> " backup which is capable of supp						
			operation at least one (1) hour in the event of a mai						
	2.6.2	(c)	-space quality" of ILS shall at least comply with cu	rrent Operational					
	[ET]		as follows :						
		Item	Airport /Runway	Operational Categor					
		2.6.2.1	SUVARNABHUMI Airport /Runway 02L	Cat II					
		2.6.2.2	SUVARNABHUMI Airport /Runway 20R	Cat II					
		"Low-pon	rered" DME transponder, shall <u>not</u> contribute to the ov	verall system error					
			$\pm 0.5 \mu s$ (≈ 75 m for " <u>ONE-WAY</u> " range error). Additionall						
		of the trar	nsponder errors, transponder location coordinate errors, p	propagation effects					
			m pulse interference effects shall not contribute more than						
		to the ove	erall system error [ANNEX 10 / Vol. I / Paragraph 3.5.4.5.2	and 3.5.4.5.1.2].					
	2.6.3	For each I	LS/DME system, the Contractor shall provide ILS (LOC a	nd GP) equipment					
	[ET]	that is ce	ertified for Facility Performance/Operational Category	not lower than					
		those state	d in [2.6.2], the brand of the ILS (LOC and GP) equipment a	nd DME equipment					
		shall be th	e same.						
		If the II S	'DME systems are the same Facility Performance/Ope	rational Catogony					
			forces the Tenderer to propose all such ILS/DME syst	5 .					
		specificati		erns in the same					
		specification							
				2 -					
				Or					

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	1					
2.6.4	The eq	uipment sha	ll be designed	d for high-relia	bility operation	n. Tenderer sl
[ET]	5.285 (2895			mance, such a		
		of the propo				
	Integrity	and/ar MTD		e an stance a canala a l	l ka avaatao th	
	250 60			equipment shal 3.12 and 3.1.5.8	-	ian the follow
		107 00.171			J •	
		C		DC	and the second states of the s	SP NTRO
F	acility Cat	Service Level	Integrity 1-(1.0×10 ^{-?})	MTBO MTBO 1000	Integrity 1-(1.0×10 ⁻¹)	MTBO MTBO 1000
	1	3	1-(1.0×10 ⁻)	MIBC 1000	1-(1.0×10 ⁻)	MIBO 1000
	IIIA	3	1-(0.5×10 ⁻⁹)	MTBO 2000	1-(0.5×10 ⁻⁹)	MTBO 2000
	IIIC	4		MTBO 4000		
				ance Category r	refers to the v	value which is
ap	olicable to	o any one lar	nding.			
2.6.5	The info	ormation of e	ach unit of LO	DC, GP and DM	E equipment s	hall be provid
[ET]				nixed with other		
	2.6.5.1			rdware assembl		
	2.6.5.2	Software r	menus and da	ta [6.4.2.2]		
	2.6.5.3	Technical	documents ar	nd test reports [12]	
2.6.6	All RF g	enerators of	LOC, GP and	DME equipme	nt shall be sy	nthesizers.
[ET]				2. 12		
2.6.7	Each ur	nit of LOC, G	P and DME e	quipment shall	be capable o	of independer
[EC]	data (ev	vents) logging	g (e.g. alarm	history), <u>not</u> d	epending on a	a connection
	any exte	ernal control	& monitoring (unit the LMM	and/or RMM c	omputer.
	2.6.7.1	All data (ev	vent) logs shal	l be marked wit	h "Date & Tim	e".
	2.6.7.2	"Date & Tim	e" of the equip	ment shall be ac	ljustable to be a	t the present tir
2.6.8	If the s	status data	of any LOC	or GP or DME	or RCMU equip	ment does r
[ET]	natively	support the '	" <u>ETHERNET</u> " fo	ormat for remote	e control & ma	onitoring purpc
	"ADAPTI	ERS" for co	nverting othe	er data format	s to the Eth	nernet shall
	provided	d.				
	Addition	ally, each ur	nit of LOC. GF	and DME equ	ipment shall	provide relev;
		a. 14		may be of dif	•	
				J and/or RMM c	1823	
	1	an samaannaan ann 25		Capping Coop		an Frankright and a state

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		1						
	2.6.9		ILS/DME system, the Contractor shall provide one (1) set of Interface					
	[EC]	Control D	Documents (ICDs) detailing data format exchanges.					
	9	Even the	ough the Tenderer accepts this condition, the task will again be					
		inspected	d by AEROTHAI during the Site Acceptance Test (SAT).					
	2.6.10	Environn	nental Conditions					
	[CO]	2.6.10.1	Indoor equipment shall be designed for continuous operation					
		[ET]	under the ambient temperature range of <u>at least</u> 0 °C to +50 °C					
			with a relative humidity of up to 60%.					
		2.6.10.2	Outdoor equipment shall be designed to be weatherproof.					
		[ET]	(The maximum relative humidity in Thailand is approximately 85%.)					
2.7	AC/DC	power line:	s, transmission lines, communication lines (including construction works)					
[CO]	2.7.1	The Cor	ntractor shall provide all AC/DC power lines, transmission lines					
	[EC]	(and mor	nitoring cables), communication lines and all related installation materials					
		(e.g. connectors, cable trays/ladders, conduits/ducts and cable ties).						
	2.7.2	All transmission lines shall be provided with RF connectors, that comply with						
	[ET]	IEC 61169 international standard or other international standard.						
		The Contractor shall also conclude the specifications of the indoor transmiss						
			transmission lines, and the RF connectors to AEROTHAI.					
		5459562355456 540,450	bugh the Tenderer accepts this condition, the task will again be					
3	U	inspected by AEROTHAI during the Site Acceptance Test (SAT).						
	2.7.3	If the installation work involves buried cables, they shall be of the "underground" typ						
	[EC]	AC power lines, transmission lines (and monitoring cables) and communication						
		shall be	separated into distinct groups, each of which shall be routed through					
		its own	"underground" conduit (such as HDPE or RSC), having sufficient inner					
		diameter	for easy installation of all associated cables.					
		Even thou	ugh the Tenderer accepts this condition, the task will again be inspected by					
	NO	AEROTHAI during the site installation.						
	2.7.4	The Cont	ractor shall carry out the underground cable work, at least, as follows :					
	[CO]	2.7.4.1	Trenching and underground cable work shall be carried out					
		[CO]	in accordance with Section 2 : Construction Works.					
		2.7.4.2	The Contractor shall take responsibilities for any damages to existing					
		[CO]	and/or nearby underground cables and utilities.					

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	2.7.5	The Cont	ractor shall	mark out all AC power lines, transmission lines (and monitoring		
	[EC]	cable) and	nd communication lines leading to the facilities, using appropriate india			
		/labels. Ad	ditionally,	route markers shall also be installed to ensure that the path		
		of the bu	ried cable i	s clearly visible.		
	0	Even thou	ugh the Tend	derer accepts this condition, the task will again be inspected by		
		AEROTHA	I during the	site installation.		
2.8	Lightnir	ng and Surge	e Protection	(LSP) System		
[CO]	A com	olete <i>"Light</i>	ning and Su	irge Protection (LSP)" system shall consist of		
	a)	Air-Termina	ition System)		
	b)	Down Cond	ductor Syste	m		
•	с)	Earth-Term	ination (Gro	unding) System including equipotential earth bonding		
	d)	Surge Prote	ective Devic	te (SPD)		
	Howeve	er, a LSP sys	stem relating	g to network equipment are separately stated in APPENDIX D.		
	2.8.1		(2021)	navigation station (LOC station or GP/DME station), the		
	[ET]			narize all LPS subsystems stated in [2.8.1.1] and [2.8.1.2] into		
				on diagram, which shall include :		
		2020		otection, as designed by AEROTHAI (Section 2 : Construction		
			orks)			
				cation protection, as designed by the Tenderer		
		2.8.1.1		ower protection from LPZ 0 (Lightning Protection Zone 0)		
		[CO]	83	AEROTHAI designed the entire system, as detailed in Section 2 :		
			Hoaten 7.55	on Works. The Tenderer are required to adhere to these		
		0010	specificatio			
		2.8.1.2 [CO]		communication protection from antenna systems to		
				ission lines, the Tenderer shall design and who has become actor shall provide a LSP subsystem.		
			2.8.1.2.1	The Tenderer shall design Air-Termination System to		
			[ET]	protect roof-top structure of the building/shelter, if the		
			ננין	building/shelter is near the antenna.		
			2.8.1.2.2	The Tenderer shall design <i>"RF Transmission Line"</i> surge		
			[ET]	protective devices capable of passing DC power along		
				the transmission line, if DC power is carried through the line.		

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		[ET]	devices of	The Tenderer shall design the Earth-Termina for all antenna systems, including the field moni The Earth-Termination System for the antenna sys bonded to the Earth-Termination System of AERC with an earth-resistance <u>not</u> exceeding 5 ohms Even though the Tenderer accepts this condit will again be inspected by AEROTHAI during the sit erer shall submit a list of, at least, major subsys [2.8.1.2] to AEROTHAI. The proposed details rega	itor antenna. stems shall be DTHAI [2.8.1.1] s. ion, the task ie installation.		
				ls may be omitted.			
2.9		cal Support					
[CO]		each ILS/DME system, the Contractor shall provide technical supports as follows :					
	2.9.1	The Contr	actor shall	be responsible for all installation works.			
	[CO]						
	2.9.2		The Contractor shall by oneself set up/configure (not only assist) all units				
	[CO]		8500	em until the results of " <u>COMMISSIONING</u> " fligh			
		100 mm		eet the requirements of the Civil Aviation .	Authority of		
		Thailand (
2.10	100 200 400			e to the following products - manufacture			
[CO]			the date n	ext to the completion of the final payment c	late of each		
		E system.					
	ltem	Descriptio	n		Warranty		
	(a)	"COMPLE	<u>TE</u> " Lightnin	g and Surge Protection (LSP) System	5 years		
	(b)	LOC, GP, D	ME, RCMU, I	RSU and Network Equipment	5 years		
	(c)	Spare Part	S		5 years		
	(d)	Measuring	Instrument	s	2 years		
	(e)	Computer	S		2 years		

cations Cations

3. Specifications of Localizer

[CO] The specifications described below represent AEROTHAI's requirements. ANNEX10 / Vol. I and/or other international standard references stated at the end of each item are only for citations. The interpretations <u>may</u> not exactly match AEROTHAI's specific requirement.

2									
3.1	LOC 1	Fransmitt	er characteristics						
[C0]	3.1.1	The sys	stem shall operate with horizontal polarization in the frequency band of						
	[ET]	108 MH	MHz to 111.975 MHz, with 50 KHz spacing between channels [ANNEX10 / Vol.						
		– Paragi	raph 3.1.3.2.1 and 3.1.3.2.2].						
	3.1.2 The frequency channels among LOC, GP and DME system shall b								
	[ET]	[ANNEX	10 / Vol. I – Paragraph 3.1.6.1, Table A / p. 3-103].						
	3.1.3	Where	two radio frequency carriers "COURSE (CRS)" and "CLEARANCE (CLR)",						
	[ET]	are use	ed, the frequency tolerance of radio frequency carrier shall <u>not</u> exceed						
		± 0.002	2%						
		Addition	ally, the nominal band occupied by these carriers shall be "SYMMETRICAL"						
			the assigned frequency. With all tolerances applied, the frequency						
		20100000000000000000000000000000000000	on between the carriers shall <u>not</u> be less than 5 kHz <u>nor</u> more than 14 kHz.						
		20							
	3.1.4	[ANNEX 10 / Vol. I / Paragraph 3.1.3.2.1]. 4 The 90 and 150 Hz modulating signals (for guidance information) sh							
	[CO]	the specifications as follows :							
	[00]	3.1.4.1	The frequency tolerance of 90 or 150 Hz tones shall be within $\pm 1.5\%$						
		[ET]	[ANNEX 10 / Vol. I / Paragraph 3.1.3.5.3 b) for Facility Performance Category II]						
		3.1.4.2	The modulation depth of 90 Hz and 150 Hz tones shall be adjustable, at least,						
		[ET]	between the limits of 18% and 22% [ANNEX 10 / Vol. I / Paragraph 3.1.3.5.2].						
	3.1.5		0 Hz modulating signal (for identification) shall meet the specifications as follows :						
	[CO]	3.1.5.1	The frequency tolerance of 1020 Hz tone shall be within 1020 ± 50 Hz						
		[ET]	[ANNEX 10 / Vol. I / Paragraph 3.1.3.9.2].						
		3.1.5.2	The modulation depth of 1020 Hz tone shall be adjustable, at least, between						
		[ET]	the limits of 5% and 15% [ANNEX 10 / Vol. I / Paragraph 3.1.3.9.2].						
	3.1.6	LOC ide	intification signal.						
	[CO]	3.1.6.1	The LOC identification signal shall employ the International Morse Code						
		[ET]	and be configurable to consist of two or three letters. It shall be preceded						
			by the International Morse Code signal of the letter "I", and also meet						
			the requirements specified in [ANNEX 10 / Vol. 1 / Paragraph 3.1.3.9].						

Scope of Specifications

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The procurement of two (2) ILS/DME systems

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	3.1.6.2 [EC]	The identifications of LOC and DME shall be synchronized.
	3.1.6.3	The LOC identification signal shall be automatically suppressed, whe
1	[EC]	the system is set to bypass (testing) mode.
	3.1.6.4 [EC]	The LOC identification code shall be configurable by means of software only with no necessity for hardware settings.
		Even though the Tenderer accepts this condition, the task will again b inspected by AEROTHAI during the Factory Acceptance Test (FAT).
3.1.7	An auto	matic protection shall be applied to RF power amplifiers to prevent damage
[EC]	in the e	vent that there is a high VSWR fault at the output of RF power amplifier.
2	Even th	ough the Tenderer accepts this condition, the task will again be inspected
6	by AERC	OTHAI during the Factory Acceptance Test (FAT).
3.1.8	Line sec	tions with "Plug-in Elements".
[EC]	To facilit	ate the measurement of the in-line parameters during system maintenance
	line sea	ctions with "Plug-in Elements" shall be embedded in/inserted to each
	RF trans	mission line as follows :
	Item	Name of the Transmission Line
	(a)	Transmission Line for CRS CSB (Course Carrier Plus Sideband Signal)
	(b)	Transmission Line for CRS SBO (Course Sideband Only Signal)
	(c)	Transmission Line for CLR CSB (Clearance Carrier Plus Sideband Signal)
	(d)	Transmission Line for CLR SBO (Clearance Sideband Only Signal)
	Remark	If the transmission lines originally come with internal sensors/measuremen
	from the	e factory, "Plug-in Elements" may be exempted only if the product design
	also pro	vides an alternative mean that allows maintenance engineers to crosscheck
	the corre	ctness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " measuring instruments.
9	Even the	bugh the Tenderer accepts this condition, the task will again be inspected by
	AEROTH	AI during the site installation.
3.1.9	LOC par	ameters which affect the LOC guidance signal shall be mainly adjustable
[EC]		are. However, some parameters may be additionally adjusted by hardware
	if necess	ary.
	Addition	ally, LOC transmitter shall be configurable to operate in hot-standby
	or cold-s	standby mode.

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[CO] 3.2.1.1 To monitor basic maintenance parameters; at least, power supply volta [EC] mode of operation, aerial/standby transmitter status, transmission frequen RF transmission power and environmental sensing data. 3.2.1.2 To be used as an "Integrity Certification". The monitor, in conjunction w [EC] a built-in test unit for calibration and testing, shall guarantee its that the detection capability remains accurate and correct. The process shall be done without turning off the equipment and without interrupt the operation of the "AERIAL" transmitter. If the process is being dor an indication showing the status of "Integrity Certification" shall also informed. Image: Second	LOC		pnitoring system of LOC shall serve, at least, the following purposes.
 [EC] mode of operation, aerial/standby transmitter status, transmission frequen RF transmission power and environmental sensing data. 3.2.1.2 To be used as an "Integrity Certification". The monitor, in conjunction w a built-in test unit for calibration and testing, shall guarantee its that the detection capability remains accurate and correct. The process shall be done without turning off the equipment and without interrupt the operation of the "AERIAL" transmitter. If the process is being dor an indication showing the status of "Integrity Certification" shall also informed. Even though the Tenderer accepts this condition, the task will again inspected by AEROTHAI during the Factory Acceptance Test (FAT). 3.2.1.3 To ensure that the LOC signal is still radiated within the conditio or tolerances specified in [ANNEX 10 / Vol. I / Paragraph 3.1.3.11. the monitoring system of LOC shall provide at least the following :			
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Item Parameter (a) CRS RF Level (b) CLR RF Level (c) CRS Centerline DDM	The Processing States and States		The integral monitoring system for the " <u>AERIAL</u> " transmitter shall provide,
 (a) CRS RF Level (b) CLR RF Level (c) CRS Centerline DDM 		[ET]	at least, the critical parameters as follows :
(b) CLR RF Level (c) CRS Centerline DDM Harris			Item Parameter
(c) CRS Centerline DDM H			(a) CRS RF Level
for			(b) CLR RF Level
		0	(c) CRS Centerline DDM
(d) CRS Centerline SDM			(d) CRS Centerline SDM
(e) CRS Width DDM			(e) CRS Width DDM
(f) CLR Width DDM			(f) CLR Width DDM
(g) CRS Ident (at least status)			(g) CRS Ident (at least status)
(h) CRS & CLR Frequency Difference			(h) CRS & CLR Frequency Difference

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3.2.3	The " <u>5</u>	1	monitoring system of LOC.		
[CO]	 International approximation (2) (2) (2) (2) (2) (2) (2) (2) (2) (2)				
	[ET]	at least,	the critical parameters as follows :		
		Item	Parameter		
		(a)	CRS RF Level		
		(b)	CLR RF Level		
		(c)	CRS Centerline DDM		
		(d)	CRS Centerline SDM		
3.2.4	The " <u>N</u>	EAR-FIELD	" monitoring system of LOC.		
[CO]	3.2.4.1	The near	r-field monitoring system shall provide, at least, the critical paramete		
	[ET]	as follov	vs :		
		Item	Parameter		
		(a)	Centerline RF Level		
		(b)	Centerline DDM		
		(c)	Centerline SDM		
3.2.5	Warning	and Alarm	n Conditions		
[CO]	In this (context a	an " <u>ALARM</u> " is a notification triggered when the system operate		
[00]					
	with out-of-tolerance conditions. While, a " <u>WARNING/ALERT</u> " is a notification triggered when the system operates with abnormal status but remains within tolerance				
		10 1 N 10	tolerance is defined by a numeric range, the " <u>WARNING/ALERT</u>		
1			to as a " <u>PRE-ALARM</u> ".		
	3.2.5.1		Limits" of the monitored parameters stated in [3.2.2.1] [3.2.3.		
1	[EC]		.4.1], if exist, shall be adjustable to be equal to their respectiv		
	[20]	122 2224	nit values [ANNEX 10 / Vol. I / Paragraph 3.1.3.11.2].		
	3.2.5.2		rm Limits" of the monitored parameters stated in [3.2.2.1] [3.2.3.		
	[EC]		4.1], if exist, shall be adjustable to be equal to 75% of their respectiv		
1	[=0]		nit tolerances [ANNEX 10 / Vol. I / Attachment C /Paragraph 2.8.4.7].		
		atann an			
	3253	The mo	nitoring system shall issue a "WARNING/ALERI" or an "ALARIVI		
	3.2.5.3				
326	[EC]	both in '	"AUDIBLE" and "VISUAL" mode.		
3.2.6	[EC] The mo	both in ' nitoring sy	" <u>AUDIBLE</u> " and " <u>VISUAL</u> " mode. Instem shall be configurable to either single or dual monitor system		
3.2.6 [ET]	[EC] The mo When de	both in ' nitoring sy ual monito	" <u>AUDIBLE</u> " and " <u>VISUAL</u> " mode. Instem shall be configurable to either single or dual monitor system for system is configured, the decision logic of " <u>AND</u> " and " <u>OB</u> " mod		
	[EC] The mo When de	both in ' nitoring sy	rstem shall be configurable to either single or dual monitor system or system is configured, the decision logic of " <u>AND</u> " and " <u>OR</u> " mod		

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	3.2.7	For LOC, the maximum period allowing the system to radiate out-of-tolerance signal
	[ET]	including period(s) of zero radiation (detected by the "INTEGRAL" monitoring system),
		shall be as short as practicable, not exceed 2 seconds under any circumstances
		[ANNEX 10 / Vol. I / Paragraph 3.1.3.11.3.1 and 3.1.3.11.3.2 for Facility Performance
		Category II].
		The maximum period shall also be adjustable, at least, from 0 to 2 seconds.
		Additionally, design and operation of the monitor system shall be consistent with
		the requirement that radiation shall cease " \underline{OR} " identification and navigation components
		are removed from the carrier and a warning or alarm will be provided at
		the designated " <u>REMOTE</u> " control points in the event of failure of the monitor
		system itself [ANNEX 10 / Vol. / Paragraph 3.1.3.11.4].
	9	Even though the Tenderer accepts this condition, the task will again be inspected by
	S	AEROTHAI during the Factory Acceptance Test (FAT).
3.3	LOC a	ntenna system.
[CO]	3.3.1	The Tenderer shall design and who has become the Contractor shall provide <u>both</u>
	[ET]	the transmitting (TX) antenna system and near-field monitoring (MON) antenna system.
		The TX antenna system shall be "Log-Periodic Dipole (LPD)" type.
	[EC]	After site installation, the Contractor shall also conclude the specifications of, at least,
		the following attributes :
		(a) "Directivity" of LOC antenna elements
		(b) "Number" of LOC antenna elements
		(c) "Spacing and Height" of LOC antenna elements
		(d) "Aperture Type" of LOC antenna elements
		(e) "Gain" of LOC antenna elements and/or array
		Even though the Tenderer accepts this condition, the task will again be inspected
	No	by AEROTHAI during the Site Acceptance Test (SAT).
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Remark : 1. AEROTHAI will provide all necessary "Site Survey Reports" (APPENDIX C) of each ILS/DME system so that the Tenderer can make the design. If AEROTHAI does not specify vehicle/aircraft classes or critical/sensitive areas as described in [ANNEX 10 / Vol. | / Attachment C /Figure C-3 and Table C-1], The Tenderer shall design and who has become the Contractor shall provide "Aperture Type" of LOC antenna array with the best performance. 2. The TX antenna system shall be capable of radiating the LOC signal throughout the LOC frequency band (108 – 111.975 MHz) 3. All antenna supporters - mast and LOC elevated platform or LOC support tower, shall also be included in the antenna system (see also [2.5]). "Radiation Patterns" 3.3.2 The Tenderer shall submit the "Radiation Patterns" from the LOC antenna array [ET] (resulting from all antenna elements) and verify that the proposed LOC antenna system is suitable for the terrain and obstacle siting environment described in "Site Survey Reports" (APPENDIX C). AEROTHAI requires the simulation results only the case of flat terrain without obstacles scenario [2.3]. For each ILS/DME system, the 3dB CRS CSB horizontal beamwidth of LOC shall not subtend a horizontal angle, as opposed to the center of LOC antenna array, wider than double of the worst-case angle between obstacles and runway centerline. **RWY** Centerline Obstacle Main Lobe 0 dB e = Beamwidt 10 dE 20 dB LOC Antenna Figure 3-1a : Criteria for proposing beamwidth of LOC Page 15

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.3.3	"Structure"			
ET]	The Tenderer shall submit the "Structure" from the LOC antenna array (resulting from all antenna elements), from both cases that siting environment are and are not taken into account, for AEROTHAI's consideration [2.3].			
	The "Structure" shall comply with [ANNEX 10 / Vol. I / Paragraph 3.1.3.4] [ANNEX 10 / Vol. I			
3.3.4	/ Attachment C / Note to 2.1.2.5, Figure C-1 and Figure C-2] and [Doc 8071 / Table I-4-7]. "DDM Characteristics"			
5.5.4 [ET]	The Tenderer shall submit the "DDM Characteristics" from the LOC antenna array			
[[]	(resulting from all antenna elements), from both cases that siting environment are <u>and</u> are not taken into account, for AEROTHAI's consideration [2.3].			
	The <i>"DDM Characteristics"</i> shall comply with [ANNEX 10 / Vol. I / Paragraph 3.1.3.7.4] [ANNEX 10 / Vol. I / Attachment C / Figure C-9] and [Doc 8071 / Table I-4-7].			
3.3.5	"Coverage (Usable Distance) - Power Density or Field Strength"			
[ET]	The Tenderer shall submit the <i>"Coverage (Usable Distance)"</i> from the LOC antenna array (resulting from all antenna elements), from both cases that siting environment are <u>and</u> are not taken into account, for AEROTHAI's consideration [2.3].			
	The <i>"Coverage (Usable Distance)"</i> shall comply with [ANNEX 10 / Vol. I / Paragraph 3.1.3.3] [ANNEX 10 / Vol. I / Attachment C / Figure C-7A and C-8A] and [Doc 8071 / Table I-4-7].			
	Additionally, the power of CLR signal shall be appropriately adjusted in order that the " <u>CRS-to-CLR</u> " signal ratio within the front CRS sector shall <u>not</u> be less than 10 dB for Facility Performance Category II [ANNEX 10 / Vol. I / Paragraph 3.1.3.3.4]			
	Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the flight inspection/validation.			
3.3.6	The Contractor shall install the near-field monitoring antenna system at a specific			
[EC]	distance, which complies with the manufacturer installation manual, from the center of LOC antenna array.			
3.3.7	The Contractor shall provide Double LED obstruction lights with photo switches			
ET]	as follows :			
	Item Installation Position			
	(a) At the top of both sides of LOC antenna array (2 sets).			
	(b) At the top of near-field monitoring antenna (1 set).			
pe of	Specifications Section 1: Technical Specifications Page 16			

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	The LED obstruction light shall be weatherproof and comply with [Annex 14 /Vol. I /Paragraph 6.2.3.19, Table 6-1, Table 6-2] or other international standard for obstruction lights. Additionally, the Contractor shall also submit brand and model/type in the proposal.
3.3.8 [ET]	The Contractor shall provide marking and/or lighting, which comply to [ANNEX 14 / Vol. I / Chapter 6/ 6.2], for denoting building/shelter, antenna system and obstruction light as obstacles.
	Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the site installation.

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4. Specifications of Glide Path

[CO] The specifications described below represent AEROTHAI's requirements. ANNEX10 / Vol. | and/or other international standard references stated at the end of each item are only for citations. The interpretations may not exactly match AEROTHAI's specific requirement.

4.1.1	The system shall operate with horizontal polarization in the frequency band of				
[ET]					
	- Paragraph 3.1.5.2.1 and 3.1.5.2.2].				
4.1.2	9.				
[ET]	[ANNEX10 / Vol. I – Paragraph 3.1.6.1, Table A / p. 3-103].				
4.1.3	Where two radio frequency carriers (" <u>COURSE (CRS)</u> " and " <u>CLEARANCE (CLR)</u> ")				
[ET]	are used, the frequency tolerance of radio frequency carrier shall not exceed				
	± 0.002%				
	Additionally, the nominal band occupied by these carries shall be "SYMMETRICAL"				
	about the assigned frequency. With all tolerances applied, the frequency				
	separation between the carriers shall <u>not</u> be less than 4 kHz <u>nor</u> more than 32 kHz.				
	[ANNEX 10 / Vol. I / Paragraph 3.1.5.2.1].				
4.1.4	The 90 and 150 Hz modulating signals (for guidance information) shall meet				
[CO]	the specifications as follows :				
	4.1.4.1 The frequency tolerance of 90 or 150 Hz tones shall be within \pm 1.5 %				
	[ET] [ANNEX 10 / Vol. I / Paragraph 3.1.5.5.2 b) for Facility Performance Category II]				
	4.1.4.2 The modulation depth of 90 and 150 Hz tones shall be adjustable,				
	[ET] at least, between the limits of 37.5 % and 42.5 % [ANNEX 10 / Vol. 1 / Paragraph 3.1.5.5.1].				
4.1.5	An automatic protection shall be applied to RF power amplifiers to prevent damage				
[EC]	in the event that there is a high VSWR fault at the output of RF power amplifier.				
0	Even though the Tenderer accepts this condition, the task will again be inspected				
No	by AEROTHAI during the Factory Acceptance Test (FAT).				
4.1.6					
[EC]					
	fed into the GP antenna, GP Antenna Phaser shall be embedded/inserted into				
	each RF transmission line output of <i>"The GP Distribution Unit (DU)"</i> , at least, as follows :				
	A Resting 1. Technical Specifications Page 19				
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	AND A				

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	Item	Name of the Transmission Line	GP Ant Phas		
	(a)	Transmission Line to the Upper Antenna			
	(b)	Transmission Line to the Middle Antenna	as designe		
	(c)	Transmission Line to the Lower Antenna	1		
	Remark				
	AEROTH	HAI also accepts some certain designs that,			
	(a) No	o GP Antenna Phaser is embedded /inserted into "Tra	nsmission Line		
	to	o the Middle Antenna" [4.1.6 (b)], or			
	(b) G	² Antenna Phasers are alternatively adjusted by mean	s of software.		
		ough the Tenderer accepts this condition, the task wil HAI during the site installation.	l again be inspected		
4.1.7	Line sea	ctions with "Plug-in Elements".			
[EC]	To facili	tate the measurement of the in-line parameters durin	e system maintenand		
[=0]					
	line sections with <i>"Plug-in Elements"</i> shall be embedded/inserted into each R transmission line of [4.1.6].				
	transmi				
	Remark : If the transmission lines originally come with internal sensors/measuremen				
	from the factory, "Plug-in Elements" may be exempted only if the product design				
	norn en	le lactory, rug-in Liements may be exempted ont	y ii the product desi		
		pvides an alternative mean that allows maintenance e			
	also pro		engineers to crossche		
	also pro	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> "	engineers to crossche measuring instruments.		
	also pro the corre Even th	ovides an alternative mean that allows maintenance e	engineers to crossche measuring instruments.		
4.1.8	also pro the corre Even th AEROTH	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will	engineers to crossche measuring instruments lagain be inspected		
	also pro the corre Even th AEROTH GP para	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall	engineers to crossche measuring instruments. Lagain be inspected be mainly adjustab		
4.1.8 [EC]	also pro the corre Even th AEROTH GP para	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will tal during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally	engineers to crossche measuring instruments. Lagain be inspected be mainly adjustab		
	also pro the corre Even th AEROTH GP para by softw if neces	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally sary.	engineers to crossche measuring instruments. Lagain be inspected be mainly adjustab adjusted by hardwar		
	also pro the corre Even th AEROTH GP para by softw if neces Addition	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally sary. nally, GP transmitter shall be configurable to op	engineers to crossche measuring instruments. Lagain be inspected be mainly adjustab adjusted by hardwar		
	also pro the corre Even th AEROTH GP para by softw if neces Addition	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally sary.	engineers to crossche measuring instruments Lagain be inspected be mainly adjustab adjusted by hardwar		
	also pro the corre Even th AEROTH GP para by softw if neces Addition	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally sary. nally, GP transmitter shall be configurable to op	engineers to crossche measuring instruments Lagain be inspected be mainly adjustab adjusted by hardwar		
	also pro the corre Even th AEROTH GP para by softw if neces Addition	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally sary. nally, GP transmitter shall be configurable to op	engineers to crossche measuring instruments Lagain be inspected be mainly adjustab adjusted by hardwar		
	also pro the corre Even th AEROTH GP para by softw if neces Addition	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally sary. nally, GP transmitter shall be configurable to op	engineers to crossche measuring instruments. Lagain be inspected be mainly adjustab adjusted by hardwar		
	also pro the corre Even th AEROTH GP para by softw if neces Addition	ovides an alternative mean that allows maintenance e ectness of those <i>"in-line"</i> parameter values by " <u>EXTERNAL</u> " ough the Tenderer accepts this condition, the task will IAI during the site installation. ameters which affect the GP guidance signal shall ware. However, some parameters may be additionally sary. nally, GP transmitter shall be configurable to op	engineers to crossche measuring instruments. Lagain be inspected be mainly adjustab adjusted by hardwar		

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4.2.1	The mo	onitoring system of GP shall serve, at least, the following purposes.
[CO]	4.2.1.1	To monitor basic maintenance parameters; at least, power supply voltage,
	[EC]	mode of operation, aerial/standby transmitter status, transmission frequency,
		RF transmission power and environmental sensing data.
	4.2.1.2	To be used as an "Integrity Certification". The monitor, in conjunction with
	[EC]	a built-in test unit for calibration and testing, shall guarantee itself
		that the detection capability remains accurate and correct. The process
		shall be done without turning off the equipment and without interrupting
		the operation of the "AERIAL" transmitter. If the process is being done,
		an indication showing the status of <i>"Integrity Certification"</i> shall also be informed.
		Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Factory Acceptance Test (FAT).
	4.2.1.3	To ensure that the GP signal is still radiated within the conditions
	[ET]	or tolerances specified in [ANNEX 10 /Vol. I /Paragraph 3.1.5.7.1], the monitoring
		system of GP shall provide at least the following :
		4.2.1.3.1 "INTEGRAL" monitoring system [4.2.2].
		4.2.1.3.2 " <u>STAND-BY</u> " monitoring system [4.2.3].
		4.2.1.3.3 " <u>NEAR-FIELD</u> " monitoring system [4.2.4].
	4.2.1.4	To be used as "Fault Detection". GP equipment shall be able to detect
	[EC]	and initiate an appropriate "notification" and/or "action", if any related
		abnormal condition occurs.
4.2.2	The "IN	TEGRAL" monitoring system of GP
[CO]	4.2.2.1	The integral monitoring system for the " <u>AERIAL</u> " transmitter shall provide,
	[ET]	at least, the critical parameters as follows :
		Item Parameter
		(a) CRS RF Level
		(b) CLR RF Level
		(c) Path DDM (relative to 0 DDM)
		(d) Path SDM
		(e) Width DDM
		(f) CRS & CLR Frequency Difference
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[CO]	4.2.3.1	The stand-by monitoring system for the "STAND-BY" transmitter shall provide,		
	[ET]	at least, the critical parameters as follows :		
		Item Parameter		
		(a) CRS RF Level		
		(b) CLR RF Level		
		(c) Path DDM (relative to 0 DDM)		
		(d) Path SDM		
4.2.4	The " <u>NEAR-FIELD</u> " monitoring system of GP.			
[CO]	4.2.4.1	The near-field monitoring system shall provide, at least, the critical parameters		
	[ET]	as follows :		
		Item Parameter		
		(a) Path RF Level		
		(b) Path DDM		
		(c) Path SDM		
4.2.5				
[CO]				
	with out-of-tolerance conditions. While, a " <u>WARNING/ALERT</u> " is a notification triggered			
	14.	he system operates with abnormal status but remains within tolerance.		
		that the tolerance is defined by a numeric range, the "WARNING/ALERT"		
	may be	referred to as a " <u>PRE-ALARM</u> ".		
	4.2.5.1	"Alarm Limits" of the monitored parameters stated in [4.2.2.1] [4.2.3.1]		
	[EC]	and [4.2.4.1], if exist, shall be adjustable to be equal to their respective		
		alarm limit values [ANNEX 10 / Vol. I / Paragraph 3.1.5.7.1].		
	4.2.5.2	"Pre-Alarm Limits" of the monitored parameters stated in [4.2.2.1] [4.2.3.1]		
	[EC]	and [4.2.4.1], if exist, shall be adjustable to be equal to 75% of their respective		
		alarm limit tolerances [ANNEX 10 / Vol. / Attachment C /Paragraph 2.8.4.7].		
	4.2.5.3	The monitoring system shall issue a "WARNING/ALERT" or an "ALARM",		
	[EC]	both in " <u>AUDIBLE</u> " and " <u>VISUAL</u> " mode.		
4.2.6	The monitoring system shall be configurable to either single or dual monitor system.			
[ET]	When dual monitor system is configured, the decision logic of "AND" and "OR" mode			
	shall als	o be available.		
		Ø		
		2		
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e of Sp	ecificatio	ons Section 1: Technical Specifications Page 21		

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	4.2.7	For GP, the maximum period allowing the system to radiate out-of-tolerance signa
	[ET]	including period(s) of zero radiation (detected by the "INTEGRAL" monitoring system)
		shall be as short as practicable, not exceed 1 second under any circumstance
		[ANNEX 10 / Vol. I / Paragraph 3.1.5.7.3.1, 3.1.5.7.3.2 for Facility Performance Category II].
		The maximum period shall also be adjustable, at least, from 0 to 1 seconds.
		Additionally, design and operation of the monitor system shall be consistent with
		the requirement that radiation shall cease "OR" identification and navigation component
		are removed from the carrier and a warning or alarm will be provided a
		the designated " <u>REMOTE</u> " control points in the event of failure of the monito
		system itself [ANNEX 10 / Vol. / Paragraph 3.1.5.7.4].
	0	Even though the Tenderer accepts this condition, the task will again be inspected
		by AEROTHAI during the Factory Acceptance Test (FAT).
4.3	GP an	tenna system
[CO]	4.3.1	The Tenderer shall design and who has become the Contractor shall provide both
	[ET]	the transmitting (TX) antenna system and near-field monitoring (MON) antenna system
		The TX antenna system shall be array type with clearance signals – the so called
		"Capture Effect (CE)".
	[EC]	After site installation, the Contractor shall also conclude the specifications of, at least
		the following attributes :
		(a) "Directivity" of GP antenna elements
		(b) "Number" of GP antenna elements
		(c) "Height" of GP antenna elements
		(d) "Gain" of GP antenna elements and/or array
5	9	Even though the Tenderer accepts this condition, the task will again be inspected
	No	by AEROTHAI during the Site Acceptance Test (SAT).
		Remark :
		1. AEROTHAI will provide all necessary <i>"Site Survey Reports"</i> (APPENDIX C
		of each ILS/DME system so that the Tenderer can make the design.
		2. The TX antenna system shall be capable of radiating the GP signa
		throughout the GP frequency band (328.6 – 335.4 MHz)
		3. All antenna supporters mast and GP antenna tower, shall also be included
		in the antenna system (see also [2.5]).
		An

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4.3.2	"Radiation Patterns"
[ET]	The Tenderer shall submit the "Radiation Patterns" from the GP antenna a
	(resulting from all antenna elements) and verify that the proposed GP antenna syste
	suitable for the terrain and obstacle siting environment described in "Site Survey Repo
	(APPENDIX C). AEROTHAI requires the simulation results only the case of flat ter
	without obstacles scenario [2.3].
4.3.3	"Structure"
[ET]	The Tenderer shall submit the "Structure" from the GP antenna array (resu
	from all antenna elements), from both cases that siting environment are and
	not taken into account, for AEROTHAI's consideration [2.3].
	The <i>"Structure"</i> shall comply with [ANNEX 10 / Vol. I / Paragraph 3.1.5.4] [ANNEX 10 / V
	/ Attachment C / Note to 2.1.2.5, Figure C-1 and Figure C-2] and [Doc 8071 / Table I-4
	A Litt II. To show shall also submit a simulated TCU and/or DDH or
	Additionally, the Tenderer shall also submit a simulated TCH and/or RDH va
	(for each GP/DME station) to verify that the simulated value still meets the requirem
404	specified in [ANNEX 10 / Vol. I / Paragraph 3.1.5.1.4, 3.1.5.1.5 or 3.1.5.6].
4.3.4	<i>"DDM Characteristics"</i> The Tenderer shall submit the <i>"DDM Characteristics"</i> from the GP antenna a
[ET]	(resulting from all antenna elements), from both cases that siting environm
	are and are not taken into account, for AEROTHAI's consideration [2.3].
	The "DDM Characteristics" shall comply with [ANNEX 10 / Vol. / Paragraph 3.1.5
	3.1.5.6 and Attachment C / Figure C-11] and [Doc 8071 / Vol. I / Table I-4-8]
4.3.5	"Coverage (Usable Distance) - Power Density or Field Strength"
[ET]	The Tenderer shall submit the "Coverage (Usable Distance)" from the GP antenna a
	(resulting from all antenna elements), from both cases that siting environm
	are <u>and</u> are not taken into account, for AEROTHAI's consideration [2.3].
	The "Coverage (Usable Distance)" shall comply with [ANNEX 10 / Vol. I / Paragraph 3.1
	[ANNEX 10 / Vol. I / Attachment C / Figure C-10] and [Doc 8071 / Table I-4-8].
4.3.6	The Contractor shall install the near-field monitoring antenna system at a spe
[EC]	distance, which complies with the manufacturer installation manual, from the ce
	of GP antenna tower.
	R.

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4.3.7 [ET]	The Contractor shall provide Double LED obstruction lights <u>with</u> photo switches as follows :			
	Item Installation Position			
	(a) At the top of GP antenna tower (1 set).			
	(b) At the top of near-field monitoring antenna (1 set).			
	The LED obstruction light shall be weatherproof and comply with [Annex 14 /Vol. I			
	/Paragraph 6.2.3.19, Table 6-1, Table 6-2] or other international standard fo			
	obstruction lights. Additionally, the Contractor shall also submit brand and model/type			
	in the proposal.			
4.3.8	The Contractor shall provide marking and/or lighting, which comply to [ANNEX 14 / Vol.			
[ET]	/ Chapter 6/ 6.2], for denoting building/shelter, antenna system and obstruction ligh			
	as obstacles.			
9	Even though the Tenderer accepts this condition, the task will again be inspected b			
NR.	AEROTHAI during the site installation.			

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5. Specifications of DME/N

[CO] The specifications described below represent AEROTHAI's requirements. ANNEX10 / Vol. I and/or other international standard references stated at the end of each item are only for citations. The interpretations may not exactly match AEROTHAI's specific requirement.

5.1	DME Transponder characteristics							
[CO]	5.1.1	The system shall operate with vertical polarization in the frequency band of						
	[ET]		MHz to 1215 MHz, with 1 MHz spacing between channels [ANNEX10 / Vol. I -					
			ph 3.5.3.2].					
	5.1.2	The frequency channels among LOC, GP and DME system shall be correlate						
	[ET]	Addition	nally, the interrogation and reply frequencies of DME shall also be paired.					
		[ANNEX	(10 / Vol. I – Paragraph 3.1.6.1, 3.5.3.3.4, Table A / p. 3-103].					
	5.1.3	<u>Both</u> th	\underline{Both} the operating frequency of the reply signal \underline{and} the center frequency of					
	[ET]	eiver shall <u>not</u> vary more than \pm 0.002 % from their assigned frequencies						
	[ANNEX 10 / Vol. I / Paragraph 3.5.4.1.2 and 3.5.4.2.2].							
	5.1.4	For "lo	w-powered" DME, in case that interrogation pulse pairs with correct spacing					
	[ET]	and no	ominal frequency trigger the transponder to reply with an efficiency					
		of at le	of at least 70%, the minimum peak power density at the "TRANSPONDER ANTENNA"					
		d for such triggering shall be <u>not</u> more than -93 \pm 1 dBW/m ² [ANNEX 10						
		/ Vol. I / Paragraph 3.5.4.2.3.1, 3.5.4.2.3.2, 3.5.4.2.3.5 and 3.5.4.2.3.6].						
		example of unit conversion, the value of -93 ± 1 dBW/m ² is approximately						
		-72.45±	a 1 dBm, where DME antenna gain and cable loss are assumed to be 14 dB					
		and -2 dB respectively. However, if there is any additional attenuation embedded in						
U		the equipment, the attenuation value shall also be reported to AEROTHAI.						
	2	Even th	nough the Tenderer accepts this condition, the task will again be inspected					
		by AERC	OTHAI during the Factory Acceptance Test (FAT).					
	5.1.5	5 For each incoming interrogation frequency drift, the bandwidth of th						
	[CO]	D] shall meet the requirements as specified in the following :						
		5.1.5.1	Inside the frequency range of $f\pm 100$ KHz from the center frequency					
		[ET]	of interrogation signal, the transponder sensitivity shall not deteriorate					
			by more than 3 dB [ANNEX 10 / Vol. I / Paragraph 3.5.4.2.6.1].					
		5.1.5.2	Outside the frequency range of $f\pm 900$ KHz from the center frequency					
		[ET]	of interrogation signal, the interrogation signal shall not trigger the transponder					
			[ANNEX 10 / Vol. / Paragraph 3.5.4.2.6.5].					

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5.1.6		THAI during the Factory Acceptance Test (FAT). /N-Decoder Rejection, an interrogation pulse pair with a spacing of $\pm 2~\mu s$,
[ET]		
5.1.7		from the nominal value, shall be rejected. [ANNEX 10 / Vol. I / Paragraph 3.5.4.3.3].
[ET]	T	ape shall meet the requirements of [ANNEX 10 / Vol. I / Paragraph 3.5.4.1.3].
[[]]		Pulse rise time (between 10% to 90% of the leading edge) $\leq 3 \mu s$
		Pulse decay time (between 90% to 10% of trailing edge) $\approx 2.5 \mu s$, but $\leq 3.5 \mu s$
		Pulse duration (between 50% of the leading, and trailing edges) $3.5\pm0.5~\mu s$
		The instantaneous amplitude of the pulse shall <u>not</u> , at any instant between the point of the leading edge which is 95% of the maximum amplitude and the point of the trailing edge which is 95% of the maximum amplitude, fall below a value which is 95% of the maximum amplitude of the pulse.
5.1.8	Pulse pa	ir shall meet the requirements of [ANNEX 10 / Vol. I / Paragraph 3.5.4.1.4.3
[ET]	and 3.5.4	1.1.5.4].
	5.1.8.1	Pulse pair spacing shall be <u>not</u> exceed (12.00 \pm 0.10) μs .
		The peak power of the constituent pulses of any transponder pulse pair shall <u>not</u> differ by more than one (1) dB.
5.1.9	-	delay, the interval between 50% amplitude of the leading edge
[ET]	typically from the distance	nterrogation pulse <u>and</u> that of the corresponding reply pulse, shall be 50 μ s for X-channel, and shall also be decreasingly adjustable, at least, e nominal value in order to permit aircraft interrogators to indicate " <u>ZERO</u> " at a specific point remote from the transponder site [ANNEX 10 / Vol. I / a 3.5.4.4.1 and 3.5.4.4.3]
		bugh the Tenderer accepts this condition, the task will again be inspected THAI during the flight inspection/validation.
5.1.10	Dead tim	e and echo suppression shall be adjustable and also properly configured
[ET]		DME station, in order to prevent any undesired signal degrading m performance [ANNEX 10 / Vol. I / Paragraph 3.5.4.2.9, 3.5.4.3 and 3.5.4.6.2].
		ugh the Tenderer accepts this condition, the task will again be inspected by Al during the flight inspection/validation.
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5.1.11	The transponder shall	be capable of continuous operation at a transmission r	
[CO]		Repetition Rate") as follows :	
	5.1.11.1 The minimum [ET] and distance re	transmission rate, including randomly distributed pulse p eply pulse pair, shall not be less than <u>and</u> be close as practica xcept during identity [ANNEX 10 / Vol. I / Paragraph 3.5.4.1.5.6	
	5.1.11.2 The maximur [ET] which is higher	n transmission rate shall <u>not</u> be less than 4800 pp than the requirement recommended by ICAO at 2,700 \pm 90 pp fol. I / Paragraph 3.5.4.1.5.5].	
5.1.12	DME identification signa	al	
[CO]	[ET] and be configured and be configured by the International	ntification signal shall employ the International Morse Co urable to consist of two or three letters. It shall be preced ational Morse Code signal of the letter "I", and also me ents specified in [ANNEX 10 / Vol. I / Paragraph 3.5.3.6].	
	5.1.12.2 The DME ident	ification code shall be configurable by means of software or sity for hardware settings.	
		the Tenderer accepts this condition, the task will again AEROTHAI during the Factory Acceptance Test (FAT).	
5.1.13 [EC]		n shall be applied to RF power amplifiers to prevent dama is a high VSWR fault at the output of RF power amplifier.	
		14-1 V	
5.1.14 [ET]	so that <i>"the peak output</i> without turning off th	<i>t power</i> " can be measured by an external measuring instrume ne equipment and without interrupting the operation	
5.1.15 [EC]	 so that <i>"the peak output power"</i> can be measured by an external measuring instrument, without turning off the equipment and without interrupting the operation of the <u>"AERIAL</u>" transmitter. 5 DME parameters which affect the DME ranging signal shall be mainly adjustable 		
	Additionally, DME trans or cold-standby mode.	sponder shall be configurable to operate in hot-standl	
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D] 5	5.2.1	The monitoring system of DME shall serve, at least, the following purposes.				
[[CO]	5.2.1.1	To monit	or basic maintenance parameters; at least, power supply voltage,		
		[EC]		operation, aerial/standby transmitter status, interrogation/reply frequency,		
			Effective f	Radiated Power (or at least RF transmission power) and environmental		
			sensing d	ata.		
		5.2.1.2	To be use	ed as an "Integrity Certification". The monitor, in conjunction with		
		[EC]	a built-ir	n test unit for calibration and testing, shall guarantee itself		
			that the	detection capability remains accurate and correct. The process		
			may be	done with turning off the equipment, AEROTHAI will not strictly		
			require. I	f the process is being done, an indication showing the status of		
			"Integrity	Certification" shall also be informed.		
		0	Even tho	ugh the Tenderer accepts this condition, the task will again be		
			inspected	by AEROTHAI during the Factory Acceptance Test (FAT).		
			5.2.1.2.1	Be able to generate, at least, both "in tolerance" and "out of		
			[EC]	tolerance" pulse pair spacing [5.2.1.3.1].		
			5.2.1.2.2	Be able to generate, at least, both "in tolerance" and "out of		
			[EC]	tolerance" reply delay [5.2.1.3.2].		
			[.] 5.2.1.2.3	Be able to select /adjust the deviated frequency of simulated		
			[EC]	interrogation signals, at least, \pm 100 KHz and \pm 900 KHz [5.1.5].		
			5.2.1.2.4	Be able to provide "Dynamic Range" test		
			[EC]	When the power density of the actual interrogation signals		
				at the "TRANSPONDER ANTENNA" has any value between		
				the value specified in [5.1.4] up to a maximum of -22 $\mathrm{dBW/m^2}$		
				the performance of the transponder shall be maintained		
				[ANNEX 10 / Vol. / Paragraph 3.5.4.2.3.3].		
				For an example of unit conversion, the value of -22 $\mathrm{dBW/m^2}$		
				is approximately -1.45 dBm, where DME antenna gain and cable		
				loss are assumed to be 14 dBi and -2 dB respectively. However,		
				if there is any additional attenuation embedded in the equipment,		
				the attenuation value shall also be reported to AEROTHAI.		
			5.2.1.2.5	Be able to provide "Transmission Rate" test		
			[EC]	The DME transponder shall be capable of continuous operation		
				at a transmission rate, complying with [5.1.11].		
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	5.2.1.3	⊺o ensu	re that the DME signal is still radiated within the conditio		
	[EC]	or tolera	ance specified in [ANNEX 10 / Vol. I / Paragraph 3.5.4.7.2		
		In other	words, the monitor system shall initiate an appropriat		
		"notifica	tion" and/or "action" if any related abnormal condition occur		
		as follow	's :		
		5.2.1.3.1	Spacing error of transmitted pulse pair exceeds \pm 1.0 μs		
		[EC]	[ANNEX 10 / Vol. I / Paragraph 3.5.4.7.2.4 c)].		
		5.2.1.3.2	Reply delay error exceeds $\pm 0.5 \ \mu s$ for "low-powered" DME.		
		[EC]	[ANNEX 10 / Vol. I / Paragraph 3.5.4.7.2.2 b)].		
		5.2.1.3.3	A fall of 3 dB or more in transmitted power output.		
		[EC]	[ANNEX 10 / Vol. I / Paragraph 3.5.4.7.2.4 a)].		
		5.2.1.3.4	A fall of 6 dB or more in the minimum transponder receive		
		[EC]	sensitivity provided that this is <u>not</u> due to the action of		
			the receiver automatic gain reduction circuits [ANNEX 10 / Vol. I		
			Paragraph 3.5.4.7.2.4 b)].		
	5.2.1.4	To be use	ed as "Fault Detection". DME equipment shall be able to detec		
	[EC]	and initia	ate an appropriate "notification" and/or "action", if any relate		
		abnorma	l condition occurs.		
5.2.2	Warning	and Alarm	Conditions		
[CO]	In this context, an " <u>ALARM</u> " is a notification triggered when the system operate				
[00]	with out-of-tolerance conditions. While, a " <u>WARNING/ALERT</u> " is a notification triggere				
	when the system operates with abnormal status but remains within tolerance				
	In case that the tolerance is defined by a numeric range, the "WARNING/ALERT				
	may be referred to as a " <u>PRE-ALARM</u> ".				
	5.2.2.1		Limits" of the monitored parameters stated in [5.2.1.3.1		
	[EC]		3.4], if exist, shall be adjustable to be equal to their respective		
			hit values [ANNEX 10 / Vol. I / Paragraph 3.1.5.7.1].		
	5.2.2.2		itoring system shall issue an "ALARM", both in "AUDIBLE" and		
	[EC]	" <u>VISUAL</u> "			
5.2.3					
[ET]			or system is configured, the decision logic of " <u>AND</u> " and " <u>OR</u> " mod		
[L I]					

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5.2.4	For DME, the maximum period allowing the system to radiate out-of-tolerance signal
[ET]	including period(s) of zero radiation (detected by the " <u>AERIAL</u> " monitoring system), shall be as short as practicable, not exceed 10 seconds under any circumstances [ANNEX 10 / Vol. I / Paragraph 3.5.4.7.2.5].
	The maximum period shall also be adjustable, at least, from 0 to 10 seconds.
	Additionally, design and operation of the monitor system shall be consistent with the requirement that radiation shall cease " <u>OR</u> " identification and navigation components are removed from the carrier and a warning or alarm will be provided at the designated " <u>REMOTE</u> " control points in the event of failure of the monitor system itself.
ß	Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Factory Acceptance Test (FAT).
5.3 DME	antenna system
CO] 5.3.1 [ET]	The Tenderer shall design and who has become the Contractor shall provide the transponder (XPDR) antenna system (<u>no</u> near-field monitoring antenna system for DME), The TX antenna system shall be " <i>Uni-Directional</i> " type.
[EC]	After site installation, the Contractor shall also conclude the specifications of, at least, the following attributes : (a) "Main Lobe Elevation Angle" of DME antenna (3 or 6 degree) (b) "Gain" of DME antenna
R	Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Site Acceptance Test (SAT).
	 <u>Remark</u>: 1. AEROTHAI will provide all necessary <i>"Site Survey Reports"</i> (APPENDIX C) of each ILS/DME system so that the Tenderer can make the design. 2. The XPDR antenna system shall be capable of radiating the DME signal throughout the DME frequency band (960 – 1215 MHz) 3. All antenna supporters mast and GP antenna tower, shall also be included in the antenna system (see also [2.5]).
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 4. The Contractor shall mount the XPDR antenna on the "GP Antenna Tower", at the appropriate height, but lower than the top of the "GP Antenna Tower". Therefore, double LED obstruction light with photo switch shall not be provided for DME 5. The contractor shall not provide "Lightning Rod Assembly" for the XPDR antenna. 				
2 "Radiation Patterns"				
The Tenderer shall submit the <i>"Radiation Patterns"</i> from the DME antenna and verify that the proposed DME antenna is suitable for the terrain and obstacle siting environment described in <i>"Site Survey Reports"</i> (APPENDIX C). AEROTHAI requires the simulation results only the case of flat terrain without obstacle scenario [2.3]				
Coverage - Power Density or Field Strength				
The Tenderer shall submit the <i>"Coverage"</i> from the DME antenna, from both cases that siting environment are <u>and</u> are not taken into account, for AEROTHAI considerations [2.3]. The <i>"Coverage"</i> shall comply with [ANNEX 10 / Vol. I / Paragraph 3.5.3.1.2.2 and 3.5.4.1.5.2] [ANNEX 10 / Vol. I / Attachment C / Figure C-20] and [Doc 8071 / Table I-3-3].				

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6. Specification of Monitoring and Control System

[CO] A complete "monitoring & control" system of each ILS/DME system shall consist of LCMU, RCMU, RSU, LMM computer and RMM computer. Each unit provides the equipment status/information, or may also provides control function of the equipment, to relevant users at the designated location. Local Control and Monitoring Unit (LCMU) 6.1 [CO] LCMU is a "BUILT-IN" unit, used to locally monitor and control the equipment. It's typically embedded into each unit of LOC, GP and DME equipment. "LCMU of LOC", "LCMU of GP" and "LCMU of DME" shall provide, at least, 6.1.1 the functions, as described in Table 6.1. [ET] Remote Control and Monitoring Unit (RCMU) 6.2 RCMU is a unit, used to remotely monitor and control the equipment. It's typically located [CO] at the technical control room of the ATC tower. "RCMU of LOC", "RCMU of GP" and "RCMU of DME" shall provide, at least, 6.2.1 [ET] the functions, as described in Table 6.1. RCMU of [6.2.1] shall be combined into the same unit - "RCMU of ILS/DME". 6.2.2 If exists, "RCMU of ILS/DME" of the same airport /runway, in the same procurement, [EC] shall also be combined into the same unit. The Contractor shall also provide a suitable-sized rack for mounting the "RCMU 6.2.3 of ILS/DME". [ET] Remote Status Unit (RSU) 6.3 RSU is a unit, used only to remotely monitor the equipment. It's typically located [CO] at the ATC room of the ATC tower. "RSU of LOC", "RSU of GP" and "RSU of DME" shall provide, at least, the 6.3.1 as described in Table 6.1. [ET] functions, RSU of [6.3.1] shall be combined into the same unit - "RSU of ILS/DME". 6.3.2 If exists, "RSU of ILS/DME" of the same airport/runway, in the same procurement, [EC] shall also be combined into the same unit.

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6.4	Local	Mainter	nance Monitoring (LMM) and Remote Maintenance Monitor	ng (RMM)			
[CO]	Computer						
	LMM computer is a unit, used to locally monitor and control the equipment. It's typically located at each LOC or GP/DME station.						
	RMM computer is a unit, used to remotely monitor and control the equipment. It's typically located at the technical control room of the ATC tower.						
	6.4.1 LMM and RMM computer shall provide, at least, the functions, as described in Table 6.1. [ET]						
-	6.4.2	LMM and RMM computer shall be a desktop computer, complying with [7.1].					
		6.4.2.1 [EC]	6.4.2.1 One (1) LMM computer shall be provided for one (1) LOC station.				
		6.4.2.2 [EC]	All equipment software for LMM and RMM shall be compa " <u>WINDOWS OS</u> ". The equipment software shall be readily in the desktop computer (including the spare computer). Additionally, the recovery CD/DVD (or any portable data storages) be provided to AEROTHAI.	istalled in			
6.5	A Run	ı way Sele	ection System (An Interlock System)				
[CO]	6.5.1 [ET]	The Co the Inte shall ra radiatio	ontractor shall provide and configure a runway selection system erface Control Documents (ICDs)), in order to ensure that only one ILS/D adiate at a time. When switching from one ILS/DME system to anoth n from both shall be suppressed for <u>not</u> less than 20 s [ANNEX 10 ph 3.1.2.7.1] and [FAA Order 6750.16E/ Chapter 1/ Paragraph 15].	ME system er system,			
		ltem	Airport /Runway	Interlock			
		(a) (b)	SUVARNABHUMI Airport /Runway 02L SUVARNABHUMI Airport /Runway 20R	1			
		<u>Remark</u> The syr system,	mbol " 🖌 " marks the places that AEROTHAI require the runway	selection			
			nough the Tenderer accepts this condition, the task will again be OTHAI during the Site Acceptance Test (SAT).	inspected			
				Ane			

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A runway selection system shall be failsafe designed - Failure of any interlock 6.5.2 component to include the interfacility connection must ensure "no change" [ET] in radiating status [FAA Order 6750.16E/ Chapter 1/ Paragraph 15/ item a (2) (c)]. When a failure occurs as prescribed above, the design of the runway selection system shall also allow AEROTHAI personnel to dismiss that faulty runway selection system, and return to the condition that both ILS/DME systems are operated independently. Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Site Acceptance Test (SAT). A runway selection system shall be a single management system which is 6.5.3 monitored and controlled by only one software program and/or only one [EC] hardware unit, so that an end-user can easily perform such runway selection. Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Site Acceptance Test (SAT).

[CO] Table 6.1 : Functions of monitoring & control unit

The symbol " \checkmark " indicates the minimum requirements of functions provided by the monitoring and control unit.

ltem	Function	LCMU	RCMU	RSU	LMM/ RMM
1	Display the operating status and/or system parameters, and generate " <u>VISUAL</u> " and " <u>AUDIBLE</u> " warning/alarm (with volume control or mute) when failure occurs. If the unit is just only " <i>Basic Status Indicator</i> ", <u>not</u> " <i>User Interface (UI) Display</i> ", It shall also provide push button for lamp test.	~	~	~	
2	Display the " <u>MAIN/STANDBY</u> " of the transmitter /transponder. The capability to select " <u>MAIN/STANDBY</u> " will <u>not</u> be strictly required.	1			
3	Select the "LOCAL/REMOTE" control, with higher priority on the "LOCAL" control.	1	-	-	-
4	Manually turn on/off and changeover the transmitter /transponder with an " <u>ON-ANT</u> "/" <u>ON-LOAD</u> " indication.	1	~	-	1
5	Bypass the monitor.	~	-	2	~
6	Reset some designated hardware and software, in order that the equipment could attempt to turn on.	~	-	-	1
7	Adjust/configure (by software) the system parameters of <u>both</u> transmitters/transponders <u>and</u> monitors, for a specific purpose.	2	-	2	~

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7. Specifications of Computer.

7.1	Deskto	p Comp	uter		
[CO]	The Co	ontractor	shall provide "ALL in One" desktop computers, including all attached		
	device	s complyi	ing with, at least, as follows :		
	7.1.1	All com	nponents including a mouse and a keyboard shall be produced from		
	[ET]	the sam	ne manufacturer with permanent "LOGO/BRAND" on products.		
	7.1.2	Process	or/Chipset		
	[ET]	7.1.2.1	The number of processing unit : Core \ge 6 cores, Thread \ge 6 threads		
		7.1.2.2	Base clock frequency ≥ 1.2 GHz		
		7.1.2.3	Maximum single-core clock frequency ≥ 4.5 GHz		
	7.1.3	RAM			
	[ET]	7.1.3.1	Technology – DDR5 or better		
		7.1.3.2	Capacity ≥ 8 GB		
	7.1.4	One (1)	Storage Drive		
	[ET]	7.1.4.1	Solid State Drive ≥ 480 GB		
	7.1.5	One (1) Optical Disc Drive			
	[ET]	T] 7.1.5.1 Internal or portable DVD-RW Drive, or better			
	7.1.6	Graphic	Controller		
	[ET]	7.1.6.1	Built-in graphic or dedicated graphic controller		
		7.1.6.2	Graphic memory (including the memory allocated from RAM) \ge 1.0 GB		
	7.1.7	One (1)	Display		
	[ET]	7.1.7.1	≥ 21.5 inches LED with resolution 1920 x 1080 pixels		
	7.1.8	Network	king		
	[ET]	7.1.8.1	Gigabit Ethernet, or better		
		7.1.8.2	Wi-Fi, at least compliant with Wi-Fi 5 (IEEE 802.11ac) 2.4 GHz / 5 GHz		
		7.1.8.3	Bluetooth		
	7.1.9	I/O Inter	face		
	[ET]	[ET] 7.1.9.1 Serial Port, or an adapter converting USB to Serial Port			
	7.1.10	One (1) Keyboard and One (1) Mouse			
	[ET]	7.1.10.1	Each key shall be permanently printed with both Thai and English		
			characters.		
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7.1.11	Operatio	on System, Drivers and Software			
[EC]	7.1.11.1	The operating system shall be "WINDOWS-BASED".			
	7.1.11.2	Operation system, drivers and software shall be readily installed in the desktop computers (including the spare computers).			
	2	Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Site Acceptance Test (SAT).			
	7.1.11.3	The recovery CD/DVD (or any portable data storage devices) and the user's license for the software shall be provided for AEROTHAI.			
7.1.12	The De	sktop Computer shall have a manufacturer branch office authorized			
[EC]	representative in Thailand.				
7.1.13	One (1) set of office table and chair shall be provided for one (1) desktop com				
[EC]	(not inc	luding the spare computers).			

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8. Specifications of Network Equipment

[CO]

All details of network equipment (microwave and peripheral devices), will be referred to the original document in APPENDIX D. However, the bill of quantities for the Network Equipment are again summarized in APPENDIX B.

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9. Intersystem Connection and Network Diagram

[CO]	This section will depict intersystem connection and network diagram among equipment, which							
	stated in [3] to [8]. The Tenderer and who has become the Contractor shall comply with							
	the re	equireme	nts, at least, as follows :					
	9.1	For ea	ch ILS/DME system, the "Intersystem Connection and Network Diagram"					
	[EC]	([Fig. 9-1	I] and APPENDIX D) shall be used as guidelines. After site installation, the Contractor					
		shall submit the revised diagram that reflects the actual installation condition to AEROTHAI.						
		<i>"Intersystem Connection and Network Diagram"</i> of the same airport/runway, in the same procurement, may be combined into the same diagram.						
	0	Even th	nough the Tenderer accepts this condition, the task will again be inspected by					
	N	AEROTHAI during the Site Acceptance Test (SAT).						
	9.2							
	[CO] in order to fulfill a complete "monitoring & control" system as follows:							
		9.2.1	A set of monitoring & control equipment [6][7].					
		[CO]						
		9.2.2	A set of network equipment (microwave and peripheral devices) [8] and APPENDIX D.					
		[CO]						
		9.2.3	A set of optical fibers, stated in Section 2 : Construction Works.					
		[CO]						
	[CO]		procurement, the Contractor shall also be responsible for configuring network tions, till the <i>"monitoring & control"</i> system is properly functioning [2.9].					

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Figure 9-1 : Intersystem connection for <u>both</u> ILS/DME system (AEROTHAI Conceptual Diagram)



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10. Requirements of Spare Parts

ine	Tenderer and who has become the Contractor shall comply with the requirements			
of spa	are parts, at least, as follows :			
10.1	For each unit of the LOC equipment, LOC spare parts shall be provided for			
[EC]	a " <u>SINGLE</u> " configuration system (a single transmitter and a single monitor), at least,			
	power supply modules, line replaceable modules (LRMs), circuit card assemblies (CCAs),			
	and any other common subsystem, such as backplanes, RF transfer switches, RF distribution			
	unit (DU), RF combining unit (CU), and LOC antenna system. When the spare parts of			
	the LOC antenna system are as follows :			
	Item Spare Parts			
	(a) Two (2) sets of Transmitting (TX) Antenna Element			
	(b) One (1) set of Near-Field Monitoring Antenna, only for the Runway 02L			
	For each unit of the LOC equipment, one (1) set of double LED obstruction light			
	with photo switch shall also be provided as the spare parts.			
10.2	For each unit of the GP equipment, GP spare parts shall be provided for			
[EC]	a " <u>SINGLE</u> " configuration system (a single transmitter and a single monitor), at least,			
	power supply modules, line replaceable modules (LRMs), circuit card assembly (CCAs),			
	and any other common subsystem, such as backplanes, RF transfer switches, RF distribution			
	unit (DU), RF combining unit (CU), and GP antenna system. When the spare parts of			
	the GP antenna system are as follows :			
	Item Spare Parts			
	(a) One (1) set of Transmitting (TX) Antenna Element			
	(b) One (1) set of Near-Field Monitoring Antenna, only for the Runway 02L			
	For each unit of the GP equipment, one (1) set of double LED obstruction light			
	with photo switch shall also be provided as the spare parts.			
10.3	For each unit of the DME equipment, DME spare parts shall be provided for			
[EC]	a " <u>SINGLE</u> " system configuration (a single transponder and a single monitor), at least,			
	power supply modules, line replaceable modules (LRMs), circuit card assemblies (CCA),			
	and any other common subsystem, such as backplanes, RF transfer switches, and			
	DME antenna system. When the spare parts of the DME antenna system are as follows :			
	Item Spare Parts			
	(a) One (1) set of DME Antenna, only for the Runway 02L			
	not including "Lightning Rod Assembly" for the XPDR antenna.			
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10.4	For this procurement, one (1) unit of "RCMU of ILS/DME" shall be provided as a spare unit				
[EC]	(see also [6.2.2]).				
	Reminded that, <i>"RCMU of ILS/DME"</i> shall be combined into the same unit.				
10.5	For this procurement, one (1) unit of "RSU of ILS/DME", shall be provided as a spare unit				
[EC]	(see also [6.3.2])				
	Reminded that, "RSU of ILS/DME" shall be combined into the same unit.				
10.6	For this procurement, one (1) unit of "the runway selection system (the interlock system)"				
[EC]	shall also be provided as a spare unit (see also [6.5.1]).				



11. Requirements of Supplements.

 and <i>"Tools & Accessories"</i>, at least, as follows : The Contractor shall submit a list of <i>"Measuring Instruments"</i> [11.2] and [11.3], suitable for system calibration and maintenance, identifying brand and model of each item. Additionally, the Contractor shall also submit certificates, test reports, operation manual and service manual, that cover all of the "<u>DELIVERED</u>" measuring instruments. The Contractor shall submit a list of <i>"Tools & Accessories"</i> [11.4], suitable for system calibration and maintenance, with no necessity to identify brand and model of each item. Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Site Acceptance Test (SAT). For each airport/runway, the Contractor shall provide only one (1) Portable Navigational 		
 11.1.1 The Contractor shall submit a list of "Measuring instruments" [11.2] and [11.3], suitable for system calibration and maintenance, identifying brand and model of each item. Additionally, the Contractor shall also submit certificates, test reports, operation manual and service manual, that cover all of the "<u>DELIVERED</u>" measuring instruments. 11.1.2 The Contractor shall submit a list of "Tools & Accessories" [11.4], suitable for get ach item. Even though the Tenderer accepts this condition, the task will again be inspected by AEROTHAI during the Site Acceptance Test (SAT). 12.2 For each airport/nurway, the Contractor shall provide only one (1) Portable Navigational Signal Analyzer (PNSA) : 11.2.1 The PNSA shall be designed for measuring critical performance parameters [EC] of at least LOC, GP and DVOR equipment. 11.2.2 In case of measuring the LOC or GP parameters, the PNSA function shall provide selectable capturing capability such as "CRS Only", "CLR Only" or "CRS & CLR", so that engineer staffs can easily diagnose which group of the ILS signals cause distortion in the ILS "<u>COMPOSITE</u>" signal. 11.2.3 The PNSA shall be designed for outdoor/field measurements with built-in [EC] battery powering, portable and compact size, weatherproof and corrosion-resistance. The Contractor shall also provide necessary accessories, at least, an antenna pole, a bag for the antenna pole and a bag for the PNSA. 11.2.4 All parameters shall be transferred to an external portable storage, via USB, [EC] in text format. 	11.1 The (contractor shall submit the basic requirements of supplements – "Measuring Instruments"
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[EC]in text format.Even though the Tenderer accepts this condition, the task will again be inspected by	10000000000000000000000000000000000000	battery powering, portable and compact size, weatherproof and corrosion-resistance. The Contractor shall also provide necessary accessories, at least, an antenna pole,
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The procurement of two (2) ILS/DME systems Suvarnabhumi International Airport /Runway 02L and Runway 20R

11.3	For eac	h airport/runway, the Contractor shall provide only one (1) set of "Measuring Instruments",					
[EC]	suitable	e for system calibration and maintenance, at least, as follows :					
	Item	List of Measuring Instruments					
	(a)	Digital Multimeter					
	(b)	Frequency Counter					
	(c)	RF Wattmeter for ILS					
	(d)	RF Power Sensor or RF Power Analyzer for DME, depending on maintenance procedures					
	(e)	Oscilloscope, with a feature which still mark and lock the cursor positions of					
	the measured signal even though the scaling is altered.						
	Therefore, when the position of 50% amplitude of the leading edge of each DME pulse						
		in [5.1.9] are zoomed and exactly known/marked, each cursor position will still					
		be locked even though the scaling is altered, in order that the value of "Reply Delay"					
		between those of DME pulses could be measured accurately. AEROTHAI also requires					
		the measurement of "time delay" parameter with a resolution of, at least, 0.01 μs .					
	Each unit of "Measuring Instruments" shall have an authorized representative in Thailan						
	Even though the Tenderer accepts this condition, the task will again be inspected						
N		HAI during the site installation.					
11.4	For eac	n airport/runway, the Contractor shall provide only one (1) set of "Tools & Accessories",					
[EC]	suitable	for system calibration and maintenance, at least, as follows :					
	ltem	List of Tools & Accessories					
	(a)	A set of watt elements, only if RF wattmeter or RF power analyzer is used					
	(b)	A directional coupler for DME, only if "BUILT-IN" coupling port is not provided [5.1.14]					
	(c)	A set of RF sampler elements for LOC and GP					
	(d)	A set of RF adapter kit					
	(e)	A set of dummy loads, only if the maintenance procedure required					
	(f)	A set of extension cards and/or cables, only if the maintenance procedure required					
	(g)	A set of test cables with specific electrical length, only if the maintenance procedure required					
	(h)	A set of tuning tools, only if the maintenance procedure required					
	(i)	A set of attenuation kit, only if the maintenance procedure required					
0	Even though the Tenderer accepts this condition, the task will again be inspected by						
(9)	Even tr	lough the renderer accepts this condition, the task witt again be inspected by					

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12. Requirements of Technical Documents and Test Reports

The	Contracto	or shall provide documents as follows :				
(All c	locumen	ts relating to network equipment (APPENDIX D) are also included herein.)				
12.1	After completion of factory acceptance test (FAT), the "FAT Report" shall be provided for					
[EC]	each IL	S/DME system :				
	12.1.1	One (1) original.				
	12.1.2	Two (2) sets of hard copy.				
	12.1.3	One (1) set of soft copy.				
	Even t	hough the Tenderer accepts this condition, the task will again be inspected b				
S	AEROT	HAI after completion of factory acceptance test (FAT).				
12.2	Before	site installation, the related "Equipment Manual" containing all information				
[EC]	about	installation, operation and maintenance procedure, shall be provided for each uni				
	of LOC	, GP, DME, RCMU, RSU and Network Equipment.				
	12.2.1	Two (2) sets of hard copy.				
	12.2.2	One (1) set of soft copy.				
2	Even t	hough the Tenderer accepts this condition, the task will again be inspected by				
	AEROTHAI before site installation.					
12.3	Before	Before site installation, the related "Assembly Drawings" and "Schematic Diagrams"				
[EC]	c] shall be provided for each unit of LOC, GP, DME, RCMU, RSU and Network Equipm					
	12.3.1	Two (2) sets of hard copy.				
	12.3.2	One (1) set of soft copy.				
2	Even t	hough the Tenderer accepts this condition, the task will again be inspected by				
	AEROTHAI before site installation.					
12.4	After completion of site acceptance test (SAT) and commissioning flight inspect					
[EC]	the "S	AT Report" shall be provided for <u>each</u> ILS/DME system, The document shall				
	include	information about LOC, GP, DME, RCMU, RSU and Network Equipment.				
	12.4.1	One (1) original.				
	12.4.2	Two (2) sets of hard copy.				
	12.4.3	One (1) set of soft copy.				
2	Even t	hough the Tenderer accepts this condition, the task will again be inspected by				
NO	AEROTI	HAI after completion of site acceptance test (SAT).				

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Scope of Specifications

Section 1: Technical Specifications

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The procurement of two (2) ILS/DME systems

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APPENDIX A

List of ILS/DME Systems

		A transmit	Designal	Frequency /Channel		
ltem	Airport / Approach Runway	Airport Regional Operator Control Center	LOC	GP	DME	
				(MHz)	(MHz)	(CH.)
1.	Suvarnabhumi International Airport /RWY 02L	AOT	CENTER	108.7	330.5	24X
2.	Suvarnabhumi International Airport /RWY 20R	AUT	CENTER	111.7	333.5	54X



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APPENDIX B

Supporting Documents and Bill of Quantities

The Tenderer shall submit the supporting documents as stated in Table B.1, and who has become the Contractor shall provide the quantities as stated in Table B.2 :

Table B.1	: S	upporting	Documents
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ltem	Topic		Quantity	Remark
1	A su	mmary of the complete LSP system, which is "ONE"	2	See [2.8.1]
	interd	connection diagram for each type of navigation station	(1 for LOC	
	(LOC	station or GP/DME station), that covers the details,	1 for GP/DME)	
	at lea	ast, as follows :		
	a)	AC power protection, as designed by AEROTHAI		
	b)	Telecommunication protection, as designed by the Tenderer		
		(not stated in Section 2 : Construction Works)		[2.8.1.2]
		Additionally, list of major LSP subsystems and/or devices		
		for telecommunication protection shall also be submitted.		
2	A des	ign of navigation equipment, at least, as follows :		See [2.3]
	a)	Brand and models of the LOC, GP and DME equipment	3	[2.4]
	b)	Specifications of LOC antenna supporter and GP antenna tower	2	[2.5]
	c)	Simulation results of [2.3],	6	[2.6.4]
	d)	Performance report of LOC and GP equipment,	2	
		such as "INTEGRITY" and/or "MTBO"		
	e)	Frangibility test or evaluation report of sample, related to	2	
		LOC antenna supporter and GP antenna tower, by a method		
		complying with Frangibility Standard [Doc 9157, Part 6		
		– Frangibility / Chapter 5 or 6]. However, the Tenderer		
		shall not submit the report related to GP/DME shelters,		
		because AEROTHAI has already designed and evaluated		
		GP/DME shelter, based on such Frangibility Standard.		
	Then	the justification for the Tenderer's chosen specifications		
	shall	also be submitted.		

Scope of Specifications

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tem	Descriptions	Quantity	Remark
1	LOC Shelter	2	See [2.4].
	GP/DME Shelter	2	The quantities are already stated in BOQ
			of section 2: Construction Works
2	LOC TX Antenna Supporter	2	See [2.5].
	LOC NF Antenna Supporter	2	
	GP TX Antenna Tower	2	
	GP NF Antenna Supporter	2	
	DME XPDR Antenna Supporter	2	
3	A Set of AC power lines	2	See [2.7.1] for each ILS/DME system.
	A Set of Transmission lines	2	
	A Set of Communication Lines	2	
	A Set of Installation Materials	2	
4	A Complete LSP system for LOC	2	See [2.8.1.2] for each LOC or GP/DME station.
	A Complete LSP system for GP/DME	2	
5	A summary of the LOC specifications	1	See [3.3.1]
	with the key attributes		
	A summary of the GP specifications	1	See [4.3.1]
	with the key attributes		
	A summary of the DME specifications	1	See [5.3.1]
	with the key attributes		
6	LOC Equipment	2	See [3.1] and [3.2].
	GP Equipment	2	See [4.1] and [4.2].
	DME Equipment	2	See [5.1] and [5.2].
	TX Antenna System for LOC	2	See [3.3] and [10.1].
	- Elements as spare units	4 El.	
	TX Antenna System for GP	2	See [4.3] and [10.2].
	- Elements as spare units	2 El.	
	XPDR Antenna System for DME	2	See [5.3] and [10.3].
	- A whole spare unit	1	
	NF MON Antenna System for LOC	2	See [3.3] and [10.1].
	- A whole spare unit	1	9
	NF MON Antenna System for GP	2	See [4.3] and [10.2].
	- A whole spare unit	1	

Table B.2 : Bill of Quantities for ILS/DME systems and Network Equipment

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tem	Descriptions	Quantity	Remark
7	Adapter to Ethernet Data Format	as designed	See [2.6.8], only if the status data
			do <u>not</u> natively support the Ethernet
			format.
8	Double LED OBS Light for LOC	6	See [3.3.7] and [10.1],
	- A whole spare unit	2	including photo switch.
	Double LED OBS Light for GP	4	See [4.3.7] and [10.2],
	- A whole spare unit	2	including photo switch.
	Double LED OBS Light for DME	N/A	
9	RCMU of ILS/DME	1	See [6.2] and [10.4].
	- A whole spare unit	1	If exists, RCMU of ILS/DME of the same
			airport/runway, in the same procurement,
			shall also be combined into the same unit.
			The Contractor shall also provide
			a suitable-sized rack for mounting
			the "RCMU of ILS/DME".
10	RSU of ILS/DME	1	See [6.3] [6.5] and [10.5].
10	- A whole spare unit	1	If exists, RSU of ILS/DME of the same
	A whole spare and	*	airport/runway, in the same procurement,
			shall also be combined into the same unit.
11	Runway Selection System	1	The runway selection system may <u>not</u> need
**	- A whole spare unit	1	to be combined into the same unit with
	A whole spare and	-	the RSU.
12	Desktop Computer for LMM	4	See [6.4.2.1].
	- A whole spare unit	1	One (1) LMM computer shall be provided
	(with components in the remark)		for one (1) LOC station and one (1)
	Desktop Computer for RMM	1	GP/DME station
	(with components in the remark)		
			One (1) RMM computer shall be provided
			for one (1) airport
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tem	Descriptions	Quantity	Remark
			One (1) computer shall also be provided
			as a spare unit for one (1) airport.
			Related components shall be provided.
			a) Operating System with User's License
			b) Equipment Software
			c) Recovery CD/DVD/data storages
			for a) and b)
			A Set of Table and Chair [7.1.13]
13	Digital Microwave Radio ODU	8	See Appendix D /[4.1.1]
	Digital Microwave Radio IDU	8	See Appendix D /[4.2]
	Antenna Dish (Frequency Range 7-8 GHz)	8	See Appendix D /[4.1.2]
	Rectifier 48 Volt	8	See Appendix D /[4.3]
	Layer 2 Ethernet switch	8	See Appendix D /[4.4]
14	A design of "Intersystem Connection	2	See [9.1] and Appendix D / [5.1]
	and Network Diagram", including		"Intersystem Connection and Network
	frequency allocation of each microwave		Diagram" of the same airport / runway,
	routing, shall be submitted before		in the same procurement, may be combined
	site installation. After site installation		into the same diagram.
	a revised version of the the diagram		
	shall again be submitted.		
15	Spare Parts for LOC	2	See [10.1] [10.2] and [10.3].
	Spare Parts for GP	2	Spare parts shall be provided for a "SINGLE"
	Spare Parts for DME	2	configuration system, including any other
	Spare Parts for LSP	N/A	common subsystem.
			The spare parts of all antenna system
			are already included in [Item 6].
			The spare parts of the Double LED OBS Light
			are already included in [Item 8].
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tem	Descriptions	Quantity	Remark
16	Measuring Instrument	1	See [11.1.1][11.2] and [11.3]
	(with documents in the remark)		for each airport/runway.
	PNSA		Related documents shall be provided.
	Digital Multimeter		a) List of Measuring Instrument
	Frequency Counter		(Brands and models are required)
	RF Wattmeter for ILS		b) Certificates
	RF Power Sensor for DME		c) Test Reports
	(or RF Power Analyzer for DME)		d) Operation Manual
	Oscilloscope		e) Service Manual
17	Tools & Accessories	1	See [11.1.2] and [11.4]
	(with documents in the remark)		for each airport/runway.
	A Set of Watt Elements		Related documents shall be provided.
	A Directional Coupler for DME		a) List of Tools and Accessories, only if
	A Set of Sampler Elements for LOC and GP		the maintenance procedure required.
	A Set of RF Adapter Kit		(Brands and models are <u>not</u> required)
	A Set of Dummy Loads		
	A Set of Extension Cards and/or Cables		
	A Set of Test Cable		
	A Set of Tuning Tools		
	A Set of Attenuation Kit		
18	FAT Report (Original)	2	See [12.1].
	FAT Report (Hard Copy)	4	
	FAT Report (Soft Copy)	2	
19	SAT Report (Original)	2	See [12.4].
	SAT Report (Hard Copy)	4	for each ILS/DME system (including RCMU,
	SAT Report (Soft Copy)	2	RSU and Network Equipment).
20	Equipment Manual for LOC (HC)	4	See [12.2] and [2.6.9].
	Equipment Manual for GP (HC)	4	Where HC is Hard Copy and SC is Soft Copy.
	Equipment Manual for DME (HC)	4	
	Equipment Manual for RCMU/RSU (HC)	4	See [Appendix D /[6]]
	Equipment Manual for Network (HC)	3	The Equipment Manual of Network Equipment,
	ICDs for ILS/DME (HC)	4	shall be comprised of those, stated in
	Equipment Manual for LSP (HC)	Optional	[ltem 13]
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ltem	Descriptions		Quantity	Remark
	Equipment Manual for LOC	(SC)	2	
	Equipment Manual for GP	(SC)	2	
	Equipment Manual for DME	(SC)	2	
	Equipment Manual for RCMU/RSU	(SC)	2	
	Equipment Manual for Network	(SC)	3	
	ICDs for ILS/DME	(SC)	2	
	Equipment Manual for LSP	(SC)	Optional	
21	Assembly Drawings for LOC	(HC)	4	See [12.3]
	Assembly Drawings for GP	(HC)	4	Where HC is Hard Copy and SC is Soft Copy.
	Assembly Drawings for DME	(HC)	4	
	Assembly Drawings for RCMU/RSU	(HC)	4	
	Assembly Drawings for Network	(HC)	3	
	Assembly Drawings for LOC	(SC)	2	
	Assembly Drawings for GP	(SC)	2	
	Assembly Drawings for DME	(SC)	2	
	Assembly Drawings for RCMU/RSU	(SC)	2	
	Assembly Drawings for Network	(SC)	3	
22	Schematic Diagrams for LOC	(HC)	4	
	Schematic Diagrams for GP	(HC)	4	
	Schematic Diagrams for DME	(HC)	4	
	Schematic Diagrams for RCMU/RSU	(HC)	4	
	Schematic Diagrams for Network	(HC)	3	
	Schematic Diagrams for LOC	(SC)	2	
	Schematic Diagrams for GP	(SC)	2	
	Schematic Diagrams for DME	(SC)	2	
	Schematic Diagrams for RCMU/RSU	(SC)	2	
	Schematic Diagrams for Network	(SC)	3	
23	A set of tables and chairs		5	See [7.1.13]



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